LONG TERM HIGH TEMPERATURE CORROSION STUDIES OF HIGH TEMPERATURE ALLOYS IN CHLORINE CONTAMINATED ENVIRONMENTS

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LONG TERM HIGH TEMPERATURE CORROSION STUDIES OF HIGH TEMPERATURE ALLOYS

IN CHLORINE CONTAMINATED ENVIRONMENTS

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High temperature chlorine contaminated environments may be encountered in a number of modern industrial and energy conversion systems. Such environments have been shown to be extremely severe from a corrosion control standpoint. Rather little information is available on the corrosion properties of alloys in these environments, and the bulk of this information has been obtained in short term tests (24 hours or less). This paper will report the results of a series of high temperature corrosion tests performed on superalloys for periods up to 400 hours in an oxidizing environment consisting of argon containing 20% oxygen and 0.25% chlorine at 900 °C.

Introduction

High temperature chlorine containing environments are among the most severely corrosive environments which can be encountered in industrial processes. The behavior of metals and alloys in chlorinating environments is not well known in spite of the increasing number of processes which produce such environments, including the incineration of municipal wastes 1,2, the combustion of chlorine contaminated coal3, and waste heat recuperation from chemical and metallurgical processes which utilize chlorine as a reactant.4

The majority of the information which is readily available for high tempe rature chlorine accelerated corrosion is based upon the results of field tests in which the thermodynamic conditions producing the corrosion are not well controlled, or short term (50 hours or less) laboratory tests 5-9. Laboratory tests performed in pure Cl₂ or HCl are not likely to adequ ately predict materials performance in mixed industrial environments where additional oxidizing species such as oxygen and sulfur are also present. Furthermore, recent experiments have shown that the kinetics of chlorine accelerated corrosion reactions often do not follow regular linear or parabolic rate laws, so that the extrapolation of the results of short term experiments may lead to the overestimation or underestimation of the extent of corrosion which is encountered in practice. 10

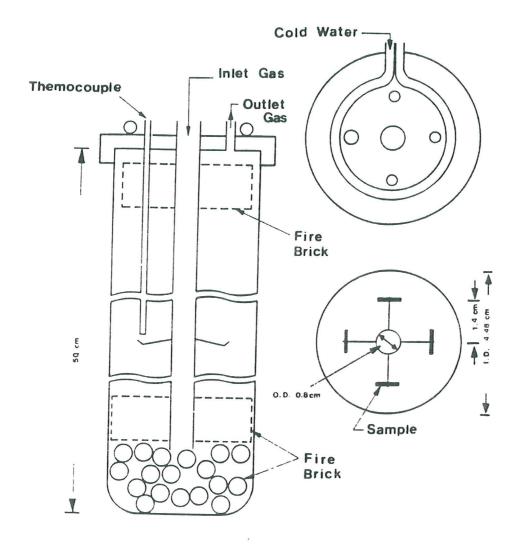
In this study, the performance of a number of commercial high temperature alloys has been studied for periods up to 400 hours in an environment consisting of argon containing 20% oxygen and 0.25% chlorine by volume. These conditions were selected after earlier experiments demonstrated that test coupons of several of the alloys could not survive 400 hours of exposure to a more corrosive environment.

Apparatus and Procedures

The apparatus used in these tests has been described elsewhere, and is shown schematically in Figure 1¹¹. It consists of a fused quartz reaction tube, and a fused quartz rack from which the specimens are suspended. Reactive gases were purified by passage through dessicant columns before being mixed and supplied to the furnace through the center of the sample rack. The gases are heated to the furnace temperature and distributed over the cross section of the reaction tube by passing them through a bed of packed, crumbled firebrick (Babcock and Wilcox K-30) at the bottom of the reaction tube.

The sample rack was designed with a knowledge of the importance of volatile corrosion product species in high temperature halogen corrosion. The corrosive gas mixture travelled upwards through the furnace with a superficial velocity of 1.5cm/sec. The samples were mounted at least 1 cm apart on a plane orthogonal to the direction of flow of the gases to prevent the inadvertant cross contamination of the specimens by gaseous or liquid corrosion products. Volatile corrosion products condensed on the water cooled cap at the top of the furnace, while condensed corrosion products either remained on the specimens or collected at the bottom of the reaction tube.

Metal specimens were in the form of one to two mm thick sheet cut into square coupons approximately one cm on a side. The compositions of the alloys included in this study are shown in Table I. The metals were supplied in the solution heat treated state with the exceptions of alloy R-41 and alloy 263 which were in the fully heat treated condition. The



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Figure 1. Schematic diagram of apparatus used for long term tests of high temperature alloys in chlorine containing environments.

TABLE I

.8Ta,.2N,.02La, .02Zr .009B,.02La others .04La 3.5Cb .006B .01Y .02* .10* .01* .10 .10 . 10 .10 .05 .04 . 08 90. .04 01 0.4* 0.4 2.2 . 3 3 0.4* 1.4 0.4 0.4 0.2 . 2 Λ1 ALLOYS (Wt.%) 0.5* 1.5* .06* ∞. Mn 0.4* 0.5* 0.5* 0.5* 0.5* *80. 0.4 0.4 0.5 . 2 Si 2.5 NOMINAL COMPOSITIONS OF 9.0 14 3 Mo 3 6 16 10 9 S 22 25 23 19 20 22 Cr 22 16 16 16 21 21 16 Bal. 20 18 00 Bal. 0.7* Bal. Bal 3* * × 19 Fe Bal. Bal. Bal. Bal. Bal. Bal. Bal. Bal. Bal. 19 33 22 20 Z. Type 310 Stainless Steel alloy C-276 INCONEL alloy No. 601 556 R-41 HAYNES alloy No. 188 800H 263 009 625 214 S alloy HAYNES alloy No. alloy CABOT alloy No. Alloy HASTELLOY HASTELLOY HASTELLOY

* Maximum

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thirteen alloys were tested in groups of four with alloy R-41 included in each group as a control to demonstrate the reproduceability of the environments.

In each test, the alloys were suspended in the furnace and the furnace was purged with argon before being heated to the test temperature. When the desired temperature was reached, the corrosive gas mixture was allowed to replace the argon and the environment was maintained for 50 hours. At the end of 50 hours, the furnace was purged with argon again, and the samples were cooled, removed from the furnace, and weighed and photographed. This procedure was repeated until the total time of exposure was 400 hours. This test procedure therefore, produces long term exposure with infrequent thermal cycling in which the temperature changes always occur in an inert environment.

Weight Change Results and External Corrosion Products

The weight change behavior exhibited by the alloys exposed in each of the four runs is illustrated in Figures 2 through 5 respectively. results show approximately linear decreases in mass for all of the alloys with the exception of alloy 214 which shows a slight increase in mass during the first 100 hours of exposure. This indicates that the principal corrosion products in this environment are volatile chloride species. After each 50 hour period of exposure, deposits of chlorides were removed from the water cooled cap at the top of the furnace. Especially large amounts of deposits were found during test number 4 where several of the alloys showed large decreases in mass. Only very small amounts of material were found at the bottom of the furnace at the conclusions of the experiments, indicating that the condensed corrosion proucts remained on the specimens during cooling. The scale of corrosion products on the specimens was very porous and friable, and some material was lost when the specimens were handled during the weighing process. Figure 6 shows the variation in the mass of the alloy R-41 specimens during each of the four tests. The reproduceability of the change in mass with time for these specimens is within 10%, and no significant effects of cross contamination could be detected.

The linear decrease in the mass of the specimens with time observed in these tests is consistent with the paralinear kinetics of corrosion observed for high temperature alloys in a more corrosive environment and for pure nickel and cobalt in the same environment in shorter term tests. In this form of corrosion, a scale of oxide corrosion products forms on the metal specimens and is subsequently attacked by chlorine leaving a porous, non-protective scale. The period during which the mass of the specimens increases with time is only a few hours long, and cannot be distinguished in long term tests such as this one.

The alloy which shows the least attack in this environment is alloy 214. This alloy contains the highest concentration of aluminum of any of the alloys in this study, and forms an oxide scale which consists primarily of aluminum oxide. INCONEL alloy No. 601, which contains a lower, but still significant concentration of aluminum, showed substantially less weight loss than alloy 600 which otherwise has a similar composition, despite the fact that it does not form a continuous alumina film. The vapor pressure of AlCl₃ in equilibrium with Al₂O₃ and the test environment is very low, and the Al₂O₃ rich scale which forms on alloy 214 appears to protect the underlying nickel and chromium rich metal from attack. The relatively poor peformance of the titanium oxide and chromium oxide forming alloys in these tests indicates that these oxides are less effective in

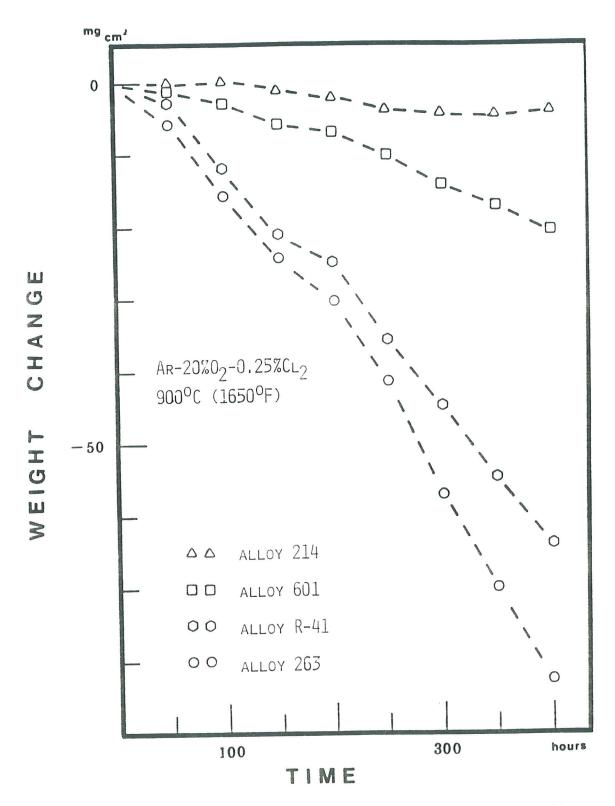


Figure 2. Weight change data from Run #1. Alloy 214, alloy 601, alloy 263, and a-loy R-41.

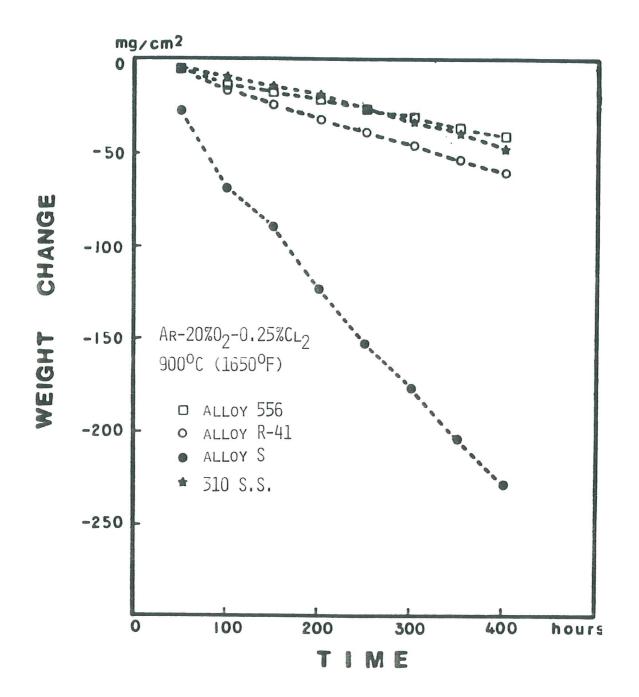


Figure 3. Weight change data for Run #2. Alloy 556, alloy S, 310 stainless steel, and alloy R-41.

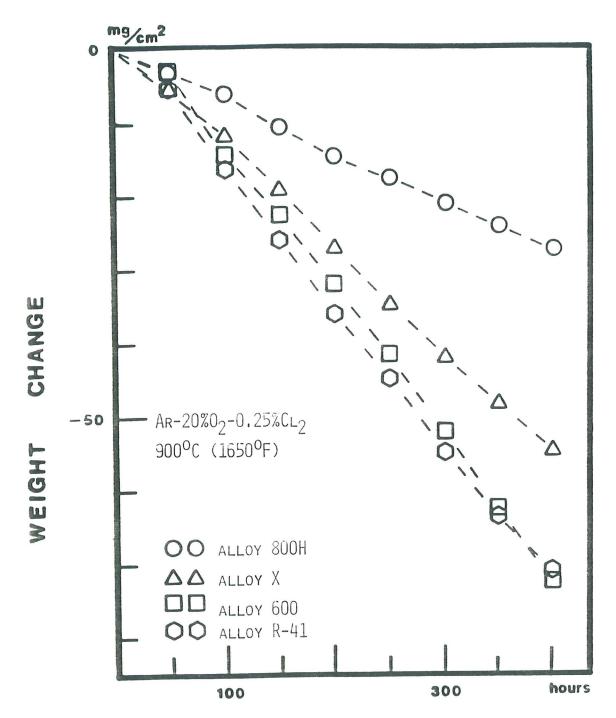


Figure 4. Weight change data for Run #3. Alloy 800H, alloy X, alloy 600, alloy R-41.

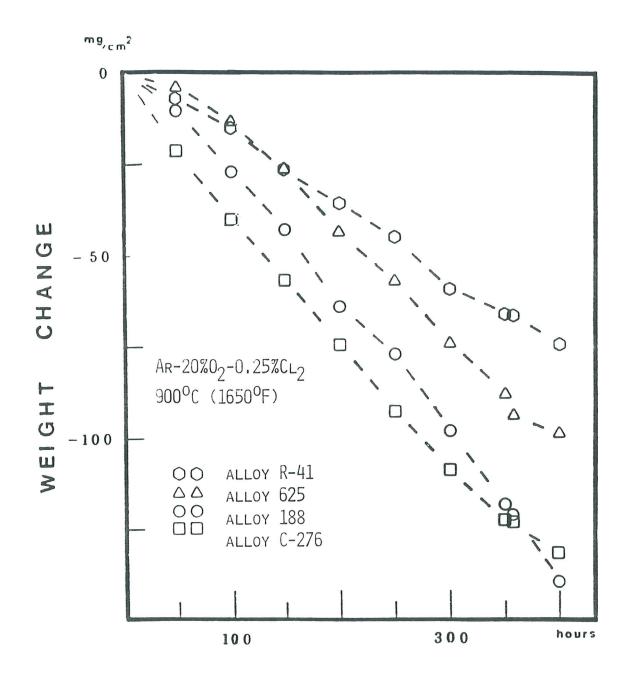


Figure 5. Weight change data from Run #4. Alloy 625, alloy 188, alloy C-276, alloy R-41.

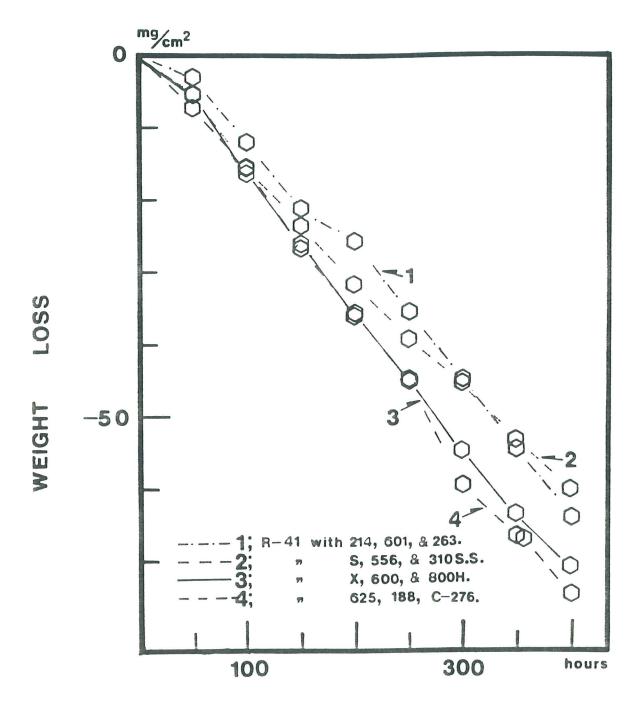


Figure 6. Weight change data for alloy R-41 specimens in all four runs.

protecting the metals from chlorine.

The second ranking materials in terms of weight loss were the iron base alloys: EAYNES alloy No. 556, alloy 800H, and 310 stainless steel. Iron forms a more stable oxide than does either nickel or cobalt, and therefore the vapor pressure of iron chloride is lower than that of nickel or cobalt chloride in environments where the oxides are thermodynamically stable. Figure 7 shows a scanning electron micrograph of the scale which formed on HAYNES alloy No. 556 along with some representative semi-quantitative analyses obtained by energy dispersive X-ray analysis. No chlorine is detected in the corrosion product scale on the specimen which consists largely of chromium oxide and contains outcropppings which are enriched in tantalum and iron. The concentrations of both nickel and cobalt in the scale are much lower than the corrosponding concentrations of these elements in the base metal.

The allcys which are attacked most severely in this environment are the cobalt and nickel based alloys which contain high concentrations of tungsten or molybdenum: notably HAYNES alloy No. 188, HASTELLOY alloy S, HASTELLOY alloy C-276, and alloy No. 625. Both tungsten and molybenum form stable oxychlorides which will have very high vapor pressures in this environment. Figure 8 shows a scanning electron micrograph of the surface of the HASTELLOY alloy S specimen at a position where a portion of the oxide scale has spalled after removal from the furnace. Again, no chlorine is detected in the corrosion product scale. The composition in region 1, where the scale had spalled to reveal the underlying metal, is very close to the nominal composition of the alloy. The oxide scale in regions 2 and 3 is enriched in chromium and iron, depleted in nickel, and contains virtualy no molybdenum.

Internal Penetration and Total Corrosive Attack

At the conclusion of the 400 hours of exposure, all of the specimens were sectioned and mounted for metallographic examination. Figure 9 shows cross sectional views of several of the alloys. Each photograph has been cropped to the original thickness of the metal specimen, so that the distance between the exterior surface of the specimen in the figure and the outside of the photograph corresponds to the metal loss by oxidat ion and volatilization. The attack occurs by uniform metal wastage, by pitting attack, and by internal attack. The specimen of alloy 625 shows a clear example of metal wastage, with only minimal internal attack very near the corroding interfaces. The specimen of INCONEL alloy 601 shows primarily pitting type attack along with somewhat more severe internal attack concentrated along the grain boundaries. The specimen of HAYNES alloy No. 188 shows very severe uniform attack as well as internal attack which penetrates through to the center of the specimen. The specimen of alloy 214 shows virtually no uniform metal wastage with all of the attack occurring along the grain boundaries.

The metallographic exmaminations revealed substantial internal attack in nearly all of the specimens. The degree of internal penetration varied from one alloy to another, and was often least severe in the alloys which showed the most severe metal loss attack. Figure 10 shows the total depth of metal affected by corrosion after 400 hours of exposure to the environment for all of the alloys included in the study. The metal wastage for each specimen is indicated by the dotted line on the bar graph, while the end of the bar inicates the average depth of penetration of the internal attack. No depth of internal attack could be determined for alloy 188 because the internal attack penetrated through the center of the

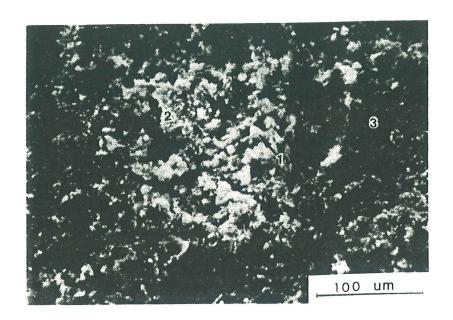


Figure 7. Corrosion product scale formed on alloy 556 after 400 hours of exposure. Semi-quantitative energy dispersive X-ray analyses.

		-						
Position	%A1	%Co	%Cr	%Fe	%Nb	%Ni	%Ta	%W
2	0.09	0.04	43.25 80.89 94.93	5.95	0.42	1.56	1.11	9.89

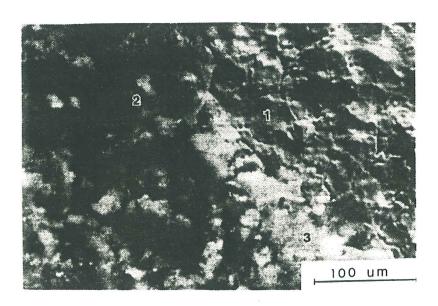


Figure 8. External surface of alloy S specimen after 400 hours of exposure showing region of spalled oxide. Semi-quantitative X-ray analyses.

Position	%A1	%Cr	%Fe	%Mo	%Ni	%Si
1	0.18	17.02	1.04	10.66	70.68	0.39
2	0.32	59.24	17.92	0.44	19.48	2.54
3	0.89	53.20	16.35	0.84	24.63	0.39

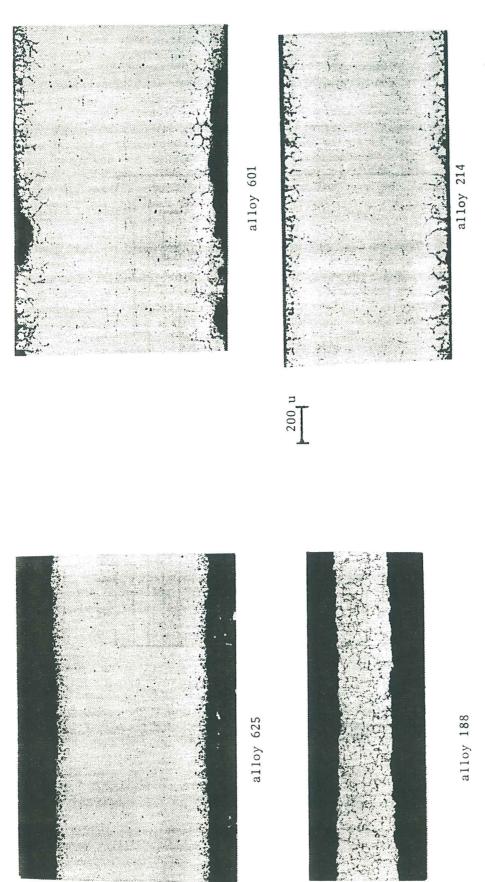


Figure 9. Optical micrographs of cross sections of alloys after 400 hours of exposure. Width of photograph corresponds to original thickness of specimen in each case.

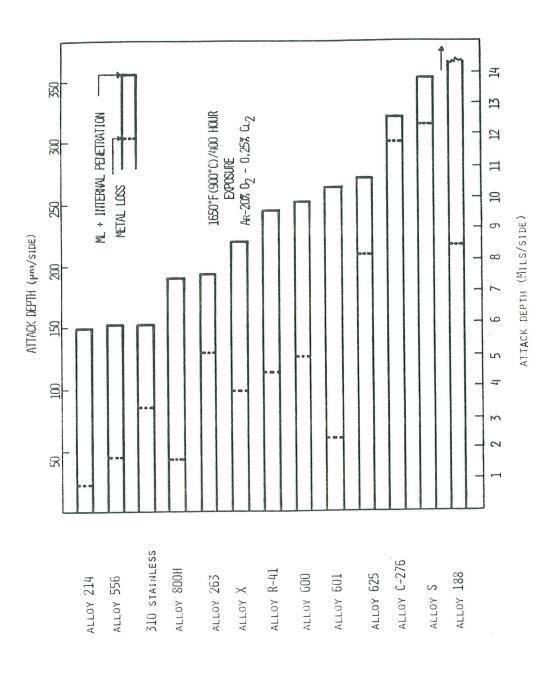


Figure 10. Metal wastage and internal penetration of specimens after 400 hours of exposure determined from optical micrographs.

specimen. CABOT alloy number 214 shows the best performance followed by the iron based alloys, and the alloys which are high in refractory metals show the most severe attack.

Figure 11 shows a scanning electron micrograph from a region of the cross section of the INCONEL alloy No 601 specimen near the surface where the internal attack can be seen clearly. The metal near the corroded surface is depleted of aluminum and chromium. The internal attack appears as open porosity, but the possibility that the attack occurs by internal formation of condensed chlorides cannot be ruled out because the samples were prepared by wet polishing, and the chlorides could have dissolved during polishing. The oxide scale on the specimen is rich in aluminum and chromium, and a small amount of chlorine can be detected in region 2 between the scale and the metal.

The scale which forms on INCONEL alloy No. 601 is not sufficiently protective to prevent the volatilization of chloride species from the specimen. CABOT alloy No. 214, which does contain sufficient aluminum to reduce the rate of volatilization from the specimen, is also subject to internal attack as shown in Figure 12. The oxide scale is rich in $Al_2\,O_3$ and the metal near the surface is depleted in aluminum. The region of internal attack is also enriched in aluminum, suggesting that both internal chloridation and internal oxidation may be occurring.

Summary and Conclusions

- 1. Long term experiments on the corrosion of high temperature alloys in oxygen-chlorine environments generally confirm the trends observed in shorter term experiments.
- 2. Corrosion occurs by two mechanisms in this environment: Metal loss by formation of volatile chlorides and oxychlorides and internal attack.
- 3. CABOT alloy No. 214, which forms an aluminum oxide protective scale, shows the least corrosion in this environment, followed by the iron based alloys HAYNES alloy No. 556, 310 stainless steel, and alloy 800H. Alloys which are high in refractory metals, including HAYNES alloy No. 188, alloy 625, and HASTELLOY alloys S and C-276, experience the most severe corrosion in this environment, presumeably because of the formation of volatile refractory metal oxychlorides.

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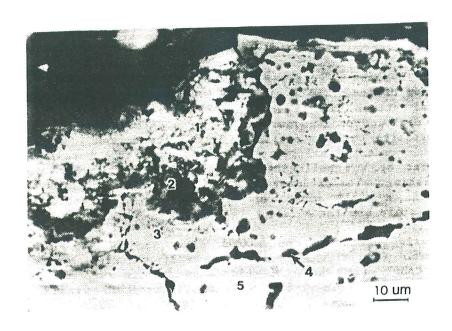


Figure 11. Scanning electron micrograph of cross section of INCONEL alloy 601 specimen near surface region showing internal attack. Semi-quantitative energy dispersive X-ray analyses

tive energy	dispersi	ve X-ray	allalyse	3			
Position	%A1	%Cr	%Fe	%Ni	%Ti	%Si	%C1
1 2 3 4 5	9.92 5.71 0 0.82 0.11	58.36 47.80 19.75 15.43 26.69	7.11 17.15 14.56 22.20 18.17	3.42 17.36 65.46 61.27 54.49	21.13 8.72 - 0.19	2.11	1.08
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Figure 12. Scanning electron micrograph of cross section of CABOT alloy No. 214 specimen near surface showing internal attack. Semi-quantative X-ray analyses.

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Position	%A1	%Cr	%Fe	%Ni	%Ti
1 2 3	70.81 32.17 3.49 0.85	13.54 20.06 17.82 21.45	1.88 2.59 3.94 3.99	13.08 44.88 74.08 73.09	0.60 0.13 0.47 0.39

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